



## Outline

- The goal and background
- Theory
- Wind tunnel overview
- Design process
- Progress
- Future work



Photo credit. National Advisory Committee for Aeronautics. Harry S. Truman Library & Museum



### The Problem

$$\rho\left(\frac{\partial u_j}{\partial t} + \frac{\partial}{\partial x_k}(u_j u_k)\right) = -\frac{\partial p}{\partial x_j} + \mu \frac{1}{Re} \frac{\partial}{\partial x_i} \left\{ \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_i}{\partial x_j}\right) - \delta_{ij} \frac{2}{3} \left(\frac{\partial u_k}{\partial x_k}\right) \right\}$$

- Often no exact solutions
- Wind tunnels are vital to the aerodynamic design process
- Houghton wants to increase experimental aerodynamic capabilities



## Approximate Theory

$$p + \frac{1}{2}\rho v^2 + \rho gh = \text{Constant}_1$$
$$\rightarrow p + \frac{1}{2}\rho v^2 = p_o$$

$$Av = Constant_2$$

### Cons. of energy

Cons. of mass

• Inefficiencies cause pressure drops

http://www.wikilectures.eu/ images/7/77/Bernoul.gif





### Closed Circuit





# Efficiency Theory

$$p + \frac{1}{2}\rho v^2 = p_o$$

$$\Delta p_{0i} = k_i \left(\frac{1}{2}\rho v^2\right)$$

- $k_i$  calculated using semiempirical equations
- Approximate stagnation pressure drop for each component



Closed Circuit 
$$\Delta p_i = k_i \left(\frac{1}{2}\rho v^2\right) \qquad \Delta p_{fan} = \sum \Delta p_i$$





### Constraints – Size of Room



#### Separation

(Photo Credit) Milton Van Dyke



### Additional Choices





### Design Process

- Script Calculates the size of each component
  - Estimate building materials
- Estimates the losses of each component
  - Estimate fan requirements











### Future work

- CFD on nozzle and corners
- Construction
- Purchasing instruments
- Future research





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